

UAV-Enabled Joint Sensing, Communication, Powering, and Backhaul Transmission in Maritime Monitoring Networks

Bohan Li¹, Member, IEEE, Jiahao Liu², Yujun Liang, Qian Li³, Haochen Liu⁴,
Yaoyuan Zhang⁵, Member, IEEE, Junsheng Mu⁶, Shahid Mumtaz⁷, Senior Member, IEEE,
and Sheng Chen⁸, Life Fellow, IEEE

Abstract—This article addresses the challenge of energy-constrained maritime monitoring networks by proposing an uncrewed aerial vehicle (UAV)-enabled integrated sensing, communication, powering and backhaul (ISCPB) transmission scheme with a tailored time-division duplex frame structure. Within each time slot (TS), the UAV sequentially implements sensing, wireless charging and uplink receiving with buoys, and lastly forward part of collected data to the central ship via backhaul links. Considering the tight coupling among these functions, we jointly optimize time allocation, UAV trajectory, UAV-buoy association, and power scheduling to maximize the performance of data collection, with the practical consideration of sea clutter effects during UAV sensing. A novel optimization framework combining alternating optimization (AO), quadratic transform (QT), and augmented first-order Taylor approximation (A-FOTE) is developed, which demonstrates good convergence behavior and robustness. Simulation results show that under sensing quality-of-service (QoS) constraint, buoys are able to achieve an average data rate over 22 bps/Hz using around 2-mW harvested power per active TS, validating the scheme's effectiveness for open-sea monitoring. Additionally, it is found

that under the influence of sea clutters, the optimal UAV trajectory always keeps a certain distance with buoys to strike a balance between sensing and other multifunctional transmissions.

Index Terms—Communication, integrated sensing, maritime monitoring network, powering and backhaul, sea clutter, uncrewed aerial vehicle (UAV).

I. INTRODUCTION

MARITIME monitoring is always the paramount approach to explore the ocean and serves as a critical foundation for marine economic development and environmental protection [1]. However, with the exponential growth in ocean data demand, the current maritime monitoring networks are facing severe challenges in terms of the achievable rates and timeliness of transmission links between monitoring platforms and data center, especially in open-sea scenarios.

To date, the acquisition of monitoring data in open sea is mainly dependent on buoy-to-ship/satellite communication links [2]. However, subject to the limited communication range and constrained mobility, the ship-assisted data collection in a large sea area is hard and time-consuming. By contrast, satellites employed with the capability of supporting widely covered communication service have dominated the data gathering method in maritime monitoring networks. For example, the world's largest ocean observation system, Global Ocean Observing System or GOOS for short, and the widely used Argo networks both utilize buoy-satellite links to obtain pelagic monitoring data [3]. In spite of the extensive coverage, the satellite-aided methods fail to provide robust and high-throughput communication services and only attain kbps-order data rates [4], owing to the fact that buoys equipped with low-cost signal processing and radio frequency (RF) units are struggled with long-distance air-space cross-medium channels. In order to overcome and compensate for the limitations of ship- and satellite-dominated monitoring data collection, the researchers have looked into the uncrewed aerial vehicle (UAV)-assisted methods.

A. Related Works

With the rapid evolution of drone technology, the modern UAV platforms, integrating high-performance baseband

Received 17 August 2025; revised 12 November 2025; accepted 27 November 2025. Date of publication 1 December 2025; date of current version 5 February 2026. This work was supported in part by the National Key Research and Development Program of China under Grant 2024YFC3109100, in part by the National Natural Science Foundation of China under Grant 62401530, in part by the Natural Science Foundation of Shandong Province under Grant ZR2024QF076, and in part by the National Defense Basic Scientific Research Program of China under Grant JCKY2024210C61424030101. (Corresponding author: Junsheng Mu.)

Bohan Li, Jiahao Liu, Yujun Liang, and Qian Li are with the Faculty of Information Science and Engineering, the Engineering Research Center of Advanced Marine Physical Instruments and Equipment (Ministry of Education), and Qingdao Key Laboratory of Optics and Optoelectronics, Ocean University of China, Qingdao 266100, China (e-mail: bohan.li@ouc.edu.cn; ljh4327@ouc.edu.cn; yujunliang@stu.ouc.edu.cn; liqian@ouc.edu.cn).

Haochen Liu is with the School of Electronics and Information, Northwestern Polytechnical University, Xi'an 710019, China (e-mail: haochenliu@nwpu.edu.cn).

Yaoyuan Zhang is with the School of Instrumentation Science and Engineering, Harbin Institute of Technology, Harbin 150001, China (e-mail: zhangyaoyuan@hit.edu.cn).

Junsheng Mu is with the School of Information and Communication Engineering, Beijing University of Posts and Telecommunications, Beijing 100876, China (e-mail: mu_junsheng@126.com).

Shahid Mumtaz is with the Department of Computer Science, Nottingham Trent University, NG1 4FQ Nottingham, U.K. (e-mail: Dr.shahid.mumtaz@ieee.org).

Sheng Chen is with the Faculty of Information Science and Engineering, Ocean University of China, Qingdao 266100, China (e-mail: sqc@ecs.soton.ac.uk).

Digital Object Identifier 10.1109/JIOT.2025.3639003

2327-4662 © 2025 IEEE. All rights reserved, including rights for text and data mining, and training of artificial intelligence and similar technologies. Personal use is permitted, but republication/redistribution requires IEEE permission.

See <https://www.ieee.org/publications/rights/index.html> for more information.

Authorized licensed use limited to: UNIVERSITY OF SOUTHAMPTON. Downloaded on February 06, 2026 at 21:23:44 UTC from IEEE Xplore. Restrictions apply.

processors and RF modules, are expected to improve the efficiency of data collection in complex maritime communication environments [5]. Ma et al. [6] leveraged the hovering UAV to collect monitoring data from sea surface sensors using nonorthogonal multiple access (NOMA) technique. Fu et al. [7] and Liu et al. [8] proposed an energy-efficient ocean data collection scheme with the aid of UAV and jointly optimized the transmit power of buoys and UAV trajectory. In the study [9], a UAV-swarm aided maritime communication network was proposed to improve connectivity and spectral efficiency based on NOMA. A multiuncrewed devices scheme consisting of UAVs and uncrewed surface vehicles (USVs) was designed in [10] to enhance the computational capacity of maritime monitoring networks, by optimizing the UAV trajectory, USV's transmit power, and computation resource. To remedy the limited coverage of terrestrial base stations (BSs), Li et al. [11] presented a hybrid satellite-UAV maritime network, in which the UAVs play the role of relay station connected with satellite and terrestrial BSs via backhaul links. However, the abovementioned papers fail to take the most unique challenges of maritime monitoring networks into consideration.

Specifically, the performance degradation caused by the frequently position variation of monitoring devices, e.g., buoys, under sea waves should be carefully addressed [12], especially when the UAV equipped with large number of antennas radiates narrow beams. To this end, the UAV may carry radar device to implement timely sensing [13] or apply beam-training toward buoys, but the former imposes extra burden on the UAV's overhead, while the latter requires buoys to be equipped with powerful baseband processors and the latency of beam-training also leads to the reduced rates [14]. Accordingly, the integrated sensing and communication (ISAC) technique is well-suited to deal with such issues, as the UAV is able to leverage one set of equipment to fulfill the flexible communication and sensing tasks [15], [16], and hence, the positions of buoys can be continuously updated, which facilitates the accurate beam matching during data transmissions. To achieve better sensing performance, the effect of sea clutters under irregular sea waves must be considered, which makes the sensing tasks in maritime networks different from that in terrestrial ones [17]. It is worth noting that the study of UAV-ISAC in maritime networks under the impact of sea clutters has not been explored in open literature.

Another challenge for maritime monitoring networks is the battery life of monitoring devices. The limited battery capacity of buoys severely restricts data offload rates and deployment longevity [18]. Unlike the cable-powered ground devices, replenishing power for buoys via cables is impractical in open sea. The UAV-assisted wireless charging technique constitutes a potentially feasible solution [19] for flexibly and sustainably charging low-cost maritime monitoring devices. Li et al. [20] and Feng et al. [21] proposed a UAV-enabled wireless charging for ground monitoring nodes and maximized the harvested energy by optimizing the UAV trajectory. A UAV-enabled simultaneous wireless information transfer and charging scheme was proposed in [22], where the UAV concurrently transmits information and implements charging

toward ground nodes based on power splitting policy, after which the ground nodes leverage the harvested power to send data to the UAV. Since the RF-based wireless charging exhibits the compatibility to various frequency bands, it can be feasibly incorporated into ISAC using the same frequency resource to form integrated sensing, communication, and powering (ISCAP) [23]. Chen et al. [24] designed a multifunctional wireless system, where BS adopts ISCAP technique to simultaneously sense target, transmit information, and charge ground nodes. A sensing-assisted scheme for robust communication and powering was proposed in [25] to improve the localization accuracy and power transfer efficiency. It can be implied that ISCAP also holds significant promise in maritime monitoring networks, as it is able to assist UAV in accurately locating monitoring devices as well as establishing stable data transmission and power transferring links with them.

B. Our Contributions

In this article, we pioneer the introduction of ISCAP in UAV-enabled maritime monitoring networks, and our novel contributions are listed as follows.

- 1) Considering an energy-limited maritime monitoring scenario, we propose a UAV-enabled integrated sensing, communication, powering and backhaul (ISCPB) transmission scheme as well as the tailor-made time-division duplex (TDD)-based frame structure, by which the UAV first senses the accurate location of buoys and then carries out power transfer and uplink transmission with buoys. In the end, to timely process latency-sensitive data, the UAV forwards part of the collected data to the ship via backhaul links. This new multifunctional transmission design improves the rates of data collection under the hostile conditions of buoy energy deficiency in open-sea scenario.
- 2) In the proposed ISCPB scheme, multiple transmission functions are tightly coupled. In particular, the sensing performance highly affects the following uplink transmission and power transferring. Also, the amount of harvested energy at buoys directly determines the uplink rates, which further influences the receivable amount of data at the ship. Therefore, we jointly optimize the time allocation among multiple functions, UAV trajectory, UAV-buoy association, and power scheduling, so as to maximize the rates of data collection. To make our proposed scheme robust in maritime scenarios, the sea clutter and buoy's displacement under wave fluctuation are practically modeled, and their impact on system performance is carefully analyzed. To the best of our knowledge, this article is the first in the field of UAV-ISAC that discusses the effect of practical sea waves.
- 3) To solve this challenging optimization problem, we design an optimization method relying on the combination of alternating optimization (AO), quadratic transform (QT), and augmented first-order Taylor approximation (A-FOTE), which exhibits good convergence behavior and robustness to various buoy monitoring scenarios. We show that during mission

TABLE I
COMPARISON BETWEEN OUR WORK AND THE RELATED EXISTING LITERATURE

	Proposed	[6, 8–10]	[7]	[11]	[15, 16]	[17]	[20, 21]	[22]	[23–25]
Maritime scenario	✓	✓	✓	✓		✓			
Practical sea wave effects (e.g., sea clutter, wave fluctuation)	✓		✓			✓			
UAV aided	✓	✓	✓	✓	✓	✓	✓	✓	
Sensing	✓				✓	✓			✓
Wireless powering	✓						✓	✓	✓
Backhaul transmission	✓			✓					
Joint optimization among multi-functional transmissions	✓			✓	✓			✓	✓
Optimization method of AO, QT and A-FOTE	✓								

period, under the constraint of sensing quality-of-service (QoS), the achievable rates of buoys per time slot (TS) are above 22 bps/Hz with self-sustaining harvested power around 2 mW per active TS, which demonstrates that the proposed scheme offers a potential solution to monitor data collection in open-sea scenario.

A specific comparison between our work and the related existing literature is concluded in Table I.

II. SYSTEM MODEL

We consider a practical scenario of data collection in open sea, where U monitoring buoys over the sea surface are dragged by the underwater anchors to avoid drifting away, and a rotary-wing UAV based in nearby ship is assigned to collect the observation data and provide power supply, from and to the energy-limited buoys, respectively. In order to deal with the buoy fluctuation caused by the sea wave, the UAV carries out the sensing tasks of localization and tracking toward the buoys, such that the antenna beams can be accurately aligned during uplink communication and wireless charging. Moreover, considering the demand of real-time data processing and the limited storage on board the UAV, a portion of the collected data are timely offloaded to the ship via wireless backhaul links during mission period.

A squared uniform plane array (UPA) with R^2 antennas is mounted on the UAV and parallel to the sea surface. Each buoy is equipped with two separate antennas for transmitting the uplink data and harvesting the energy, respectively. The ship is equipped with a single antenna for receiving the backhaul data. Let T be the UAV mission period, which is discretized into N TSs. The length of each TS $T_t = T/N$ is set to be sufficiently small, during which the location of the UAV remains approximately unchanged. Since the buoys are usually sparsely distributed in the open sea, we assume that the UAV only serves up to one buoy within each TS for the sake of power efficiency and computational complexity. Let $\alpha_u[n] \in \{0, 1\}$ be the association indicator between the u th buoy and UAV in the n th TS. If $\alpha_u[n] = 1$, the UAV executes the hybrid tasks toward the u th buoy within the n th TS, and $\alpha_u[n] = 0$ otherwise. Let $\mathcal{U} = \{1, \dots, U\}$ denote the set of buoys. We have $\sum_{u \in \mathcal{U}} \alpha_u[n] = 1$ for any TS n .

To avoid the cross-interference among the hybrid tasks of sensing, wireless charging, and communication, a TDD-like operation is adopted within every TS of the tailor-made

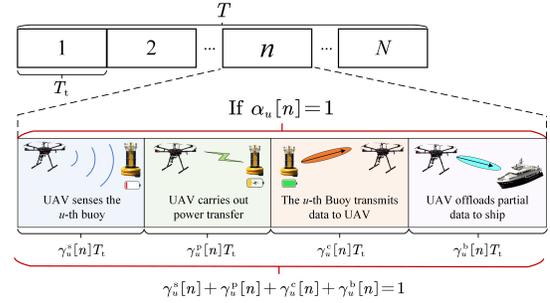


Fig. 1. Frame structure designed for UAV's ISCPB mission.

frame structure, which is shown in Fig. 1. Each TS is composed of four stages. At the first stage, relying on the prior knowledge of buoy's rough coordinates obtained by the GPS, the UAV leverages the sensing signal to precisely localize or track the position of buoy u .¹ After obtaining the accurate position information, the UAV implements the power transfer to and receives the data from buoy u , at the second and third stages, respectively. At the fourth stage, the UAV offloads a portion of the data received at the current TS to the ship via wireless backhaul links. The length of these four stages depends on the specific TS, amounting to $\gamma_u^s[n]T_t$, $\gamma_u^p[n]T_t$, $\gamma_u^c[n]T_t$, and $\gamma_u^b[n]T_t$, respectively, with $\gamma_u^s[n] + \gamma_u^p[n] + \gamma_u^c[n] + \gamma_u^b[n] = 1 \forall u, n$.

A. UAV Sensing

Assume that the UAV flies sufficiently high, such that the channels between the UAV and buoys are largely dominated by line-of-sight (LoS) links. Thus, the sensing channel of the n th TS between the UAV and the u th buoy can be expressed as

$$\mathbf{H}_u^s[n] = \sqrt{\frac{G_{\text{UAV}}^{\text{tx}} G_{\text{UAV}}^{\text{rx}} \lambda^2 \psi_u}{(4\pi)^3 d_u^4[n]}} e^{-j2\pi\tau_u[n]f_c} \times e^{j2\pi f_u^D[n]t_0} \mathbf{a}_{\text{tx}}(\theta_u[n], \phi_u[n]) \mathbf{a}_{\text{rx}}^H(\theta_u[n], \phi_u[n]) \quad (1)$$

where $G_{\text{UAV}}^{\text{tx}}$, $G_{\text{UAV}}^{\text{rx}}$, λ , ψ_u , $\tau_u[n]$, f_c , $f_u^D[n]$, and t_0 denote the antenna gains of UAV transmitter and receiver,

¹At the beginning, with the aid of GPS information, the UAV radiates sensing signal to scan the target area. Once buoy u is detected, the accurate localization is readily realized. In the following TSs related to buoy u , i.e., $\{n|\alpha_u[n] = 1\}$, the UAV can simply scan the area around the initial position to achieve tracking.

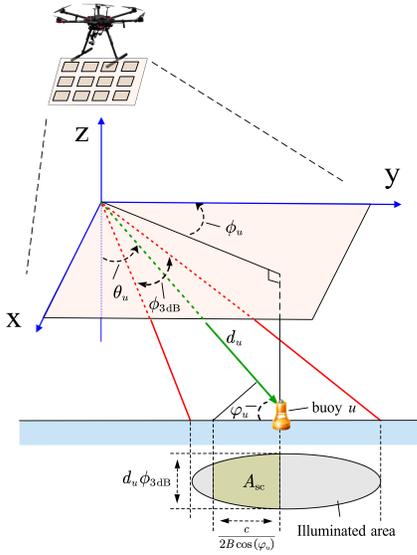


Fig. 2. Influence of sea clutter during UAV sensing.

wavelength, radar cross section (RCS), path delay, central frequency, Doppler frequency, and symbol duration, respectively, while $d_u[n]$ denotes the distance between the UAV and buoy u at TS n and $\theta_u[n]$ and $\phi_u[n]$ denote the elevation angle of departure/arrival (E-AoD/AoA) and the azimuth angle of departure/arrival (A-AoD/AoA), respectively. Both the UAV's transmit/receive array response vectors, $\mathbf{a}_{\text{tx}}(\theta_u[n], \phi_u[n])/\mathbf{a}_{\text{rx}}(\theta_u[n], \phi_u[n])$ can be expressed as

$$\mathbf{a}(\theta_u[n], \phi_u[n]) = \frac{1}{R} \left[1, \dots, e^{-\frac{j2\pi(R-1)d}{\lambda} \sin(\theta_u[n]) \cos(\phi_u[n])} \right]^T \otimes \left[1, \dots, e^{-\frac{j2\pi(R-1)d}{\lambda} \sin(\theta_u[n]) \sin(\phi_u[n])} \right]^T \quad (2)$$

where d is the antenna spacing along both the directions of x - and y -axes. Assume that the UAV flies at the constant height, and its 3-D coordinate within the n th TS is $\mathbf{c}_{\text{UAV}}[n] = [x_{\text{UAV}}[n], y_{\text{UAV}}[n], z_{\text{UAV}}[n]]^T$. The 3-D coordinates of buoys are denoted by $\mathbf{c}_u[n] = [x_u[n], y_u[n], z_u[n]]^T \forall u, n$. Then, given $d_u[n] = \|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_u[n]\|$, we have $\sin(\theta_u[n]) \cos(\phi_u[n]) = (x_{\text{UAV}}[n] - x_u[n])/d_u[n]$ and $\sin(\theta_u[n]) \sin(\phi_u[n]) = (y_{\text{UAV}}[n] - y_u[n])/d_u[n]$. Therefore, for the sensing channel, $\mathbf{a}(\theta_u[n], \phi_u[n])$ can be rewritten as $\mathbf{a}(\mathbf{c}_{\text{UAV}}[n], \mathbf{c}_u[n])$.

Let $x_u^s[n]$ be the sensing signal at the n th TS with $\mathbb{E}\{|x_u^s[n]|^2\} = 1$. Then, the UAV transmitter emits the signal $\mathbf{s}_s[n] = \sum_{u \in \mathcal{U}} \alpha_u[n] \mathbf{f}_u[n] x_u^s[n]$, where $\mathbf{f}_u[n] \in \mathbb{C}^{R^2}$ is the sensing precoder. To process the echo signal reflected by the u th buoy, the sensing combiner $\mathbf{w}_u[n] \in \mathbb{C}^{R^2}$ is applied, and the echo signal after combining is given by

$$r_u[n] = \mathbf{w}_u^H[n] (\mathbf{H}_u^s[n] + \mathbf{H}_u^{\text{sc}}[n])^H \mathbf{s}_s[n] + \mathbf{w}_u^H[n] \mathbf{n}_s[n] \quad (3)$$

where $\mathbf{n}_s[n] \in \mathbb{C}^R \sim \mathcal{CN}(\mathbf{0}, N_0 \mathbf{I}_R)$ denotes the additive white Gaussian noise (AWGN). As shown in Fig. 2, $\mathbf{H}_u^{\text{sc}}[n]$ is the

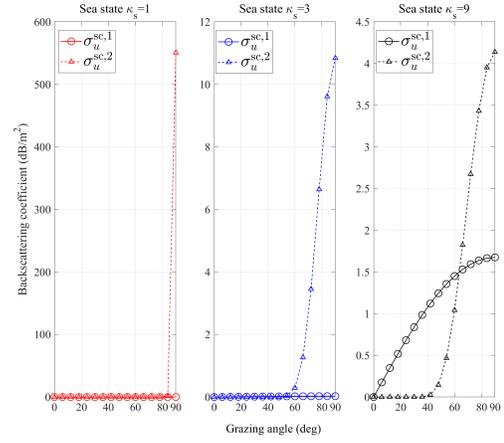


Fig. 3. Contributions of $\sigma_u^{\text{sc},1}$ and $\sigma_u^{\text{sc},2}$ to the backscattering coefficient under different sea states.

interference channel arisen from the sea clutter patch around the u th buoy, which can be expressed as

$$\begin{aligned} \mathbf{H}_u^{\text{sc}}[n] &= \sqrt{\frac{G_{\text{UAV}}^{\text{tx}} G_{\text{UAV}}^{\text{rx}} \lambda^2 \sigma_u^{\text{sc}}[n] A_u^{\text{sc}}[n]}{(4\pi)^3 d_u^4[n]}} e^{-j2\pi \tau_u^{\text{sc}}[n] f_c} \\ &\quad \times e^{j2\pi f_u^{\text{sc},D}[n] t_0} \mathbf{a}_{\text{tx}}(\mathbf{c}_{\text{UAV}}[n], \mathbf{c}_u[n]) \mathbf{a}_{\text{rx}}^H(\mathbf{c}_{\text{UAV}}[n], \mathbf{c}_u[n]) \end{aligned} \quad (4)$$

where $\sigma_u^{\text{sc}}[n]$ and $A_u^{\text{sc}}[n]$ denote the backscattering coefficient and the area of sea clutter patch, respectively. The backscattering coefficient can be formed by the widely used Morchin model [26], which is given by

$$\sigma_u^{\text{sc}}[n] = \frac{4 \times 10^{0.6(\kappa_s+1)-7} \sigma_0[n] \sin \varphi_u[n]}{\underbrace{\lambda}_{\sigma_u^{\text{sc},1}[n]} + \underbrace{\Gamma_s e^{-\tan^2(\pi/2 - \varphi_u[n]) \Gamma_s}}_{\sigma_u^{\text{sc},2}[n]}} \quad (5)$$

where the integer $0 \leq \kappa_s \leq 9$ defines the sea state, $\varphi_u[n]$ is the grazing angle, $\Gamma_s = \cot^2(2.44(\kappa_s + 1)^{1.08}/57.29)$, and $\sigma_0[n]$ is defined by

$$\sigma_0[n] = \begin{cases} \left(\frac{\varphi_u[n]}{\varphi_0} \right)^{1.9}, & \varphi_u[n] \leq \varphi_0 \\ 1, & \varphi_u[n] > \varphi_0 \end{cases} \quad (6)$$

with $\varphi_0 = \arcsin(\lambda/(0.1\pi + 0.184\pi\kappa_s))$.

Remark 1: According to (5), the backscattering coefficient is a function of sea state and grazing angle and consists of two parts. Fig. 3 shows the contributions of $\sigma_u^{\text{sc},1}$ and $\sigma_u^{\text{sc},2}$ to σ_u^{sc} under different grazing angles and sea states. It can be seen that unless the sea condition is severe, $\sigma_u^{\text{sc},2}$ dominates the backscattering coefficient. In the sequel, $\sigma_u^{\text{sc},1}$ is neglected for analysis convenience.

Moreover, the area of sea clutter patch can be derived as $A_u^{\text{sc}}[n] = (c d_u[n] \phi_{3\text{dB}})/(2B \cos(\varphi_u[n]))$, where c and B are the light speed and bandwidth, respectively, while $\phi_{3\text{dB}}$ denotes the half-power beamwidth in the azimuth plane, which is given by $\phi_{3\text{dB}} = (0.886\lambda)/(R \cdot d)$ [27]. Generally speaking, given the formulation of $\sigma_u^{\text{sc}}[n] A_u^{\text{sc}}[n]$, it can be implied that the larger the grazing angle is, the worse sea clutter may occur.

We adopt the sensing mutual information (MI) as the metric [28] to evaluate the sensing performance toward buoy u at the n th TS, which can be expressed as

$$R_u^s [n] = \gamma_u^s [n] \times \log \left(1 + \frac{\alpha_u [n] |\mathbf{w}_u^H [n] (\mathbf{H}_u^s [n])^H \mathbf{f}_u [n]|^2}{|\mathbf{w}_u^H [n] (\mathbf{H}_u^{sc} [n])^H \mathbf{f}_u [n]|^2 + N_0 \|\mathbf{w}_u [n]\|^2} \right). \quad (7)$$

Note that the sensing MI has been widely used in ISAC, since it not only has the similar properties and the same measurement unit as communication MI but also can provide a universal lower bound for the estimation-theoretic metrics, e.g., Cramér–Rao bound and mean squared error, regardless of the specific estimator [29], [30], [31]. Additionally, in our proposed frame structure, the stage of UAV sensing lasts for a while, and it is intuitive that the longer duration of sensing can lead to more accurate results of localization or tracking. Hence, the sensing MI is a better metric to take the effect of sensing time into account.

B. UAV Power Transferring

Each buoy is equipped with an energy receiver (ER), in which the received RF signals are converted into direct current (dc) signals for energy harvesting. Assume that the RF-to-dc energy conversion of the rectifier is linear,² and the maximization of the harvested dc power amounts to that of the harvested RF power. As a result, the harvested RF power at the ER of buoy u within the n th TS can be expressed as

$$P_u^p [n] = \gamma_u^p [n] \xi_u \left| \left(\check{\mathbf{h}}_u^p [n] \right)^H \mathbf{v}_u [n] \alpha_u [n] x_u^p [n] \right|^2 \quad (8)$$

where $0 \leq \xi_u \leq 1$ is the energy conversion efficiency depending on the hardware circuit of buoy u , $x_u^p [n]$ with $\mathbb{E}\{|x_u^p [n]|^2\} = 1$ is the power signal transmitted at the n th TS, and $\mathbf{v}_u [n] \in \mathbb{C}^{R^2}$ is the power transfer precoder.

It should be mentioned that although buoys are dragged by the underwater anchors, their positions will slightly change within each TS due to sea waves. Specifically, buoy u deviates from the position that was sensed by the UAV at the beginning of the TS, i.e., $\check{\mathbf{c}}_u [n] = [x_u [n] + e_x [n], y_u [n] + e_y [n], z_u [n] + e_z [n]]^T$, with $e_x [n], e_y [n], e_z [n] \in \mathcal{N}(0, \epsilon^2)$.

Remark 2: In practice, the value of $e_a [n], a \in \{x, y, z\}$ is highly related to the buoy velocity within each TS, and hence, it is reasonable to derive that $\text{Var}(e_a [n]) \triangleq \text{Var}(v_a [n]) \cdot T_i^2$, where $v_a [n]$ denotes the buoy velocity at the n th TS. Then, provided that the buoy follows the water particle motion under the deep-water and single-frequency wave and exhibits uniform velocity in every direction, $\text{Var}(v_a [n])$ can be approximately calculated as $\text{Var}(v_a [n]) \approx ((\pi H)/(T\sqrt{2}))^2$ [33], where H and T denote the wave height and wave period, respectively. Accordingly, given the statistical wave height and period under different sea

states, the variation of buoy's position deviation, i.e., ϵ^2 , can be obtained.

Then, the channel between the u th buoy and the UAV, i.e., $\check{\mathbf{h}}_u^p [n] \in \mathbb{C}^{R^2}$, is given by

$$\check{\mathbf{h}}_u^p [n] = \sqrt{\frac{G_{\text{UAV}}^{\text{rx}} G_u^{\text{tx}} \lambda^2}{(4\pi)^2 d_u^2 [n]}} e^{-j\pi\tau_u [n]f_c} e^{j2\pi f_u^p [n]t_0} \times \boldsymbol{\alpha}_{\text{tx}} (\mathbf{c}_{\text{UAV}} [n], \check{\mathbf{c}}_u [n]) \quad (9)$$

where G_u^{rx} denotes the receive antenna gain of buoy u and $d_u^2 [n] = \|\mathbf{c}_{\text{UAV}} [n] - \check{\mathbf{c}}_u [n]\|$.

C. Buoy Uplink and UAV Backhaul Model

After supplied with energy, buoys upload the monitoring data to the UAV, part of which are then relayed to the ship for timely data analysis. Let $x_u^c [n]$ and $x_u^b [n]$ be the data transmitted by buoy u and the UAV at the n th TS, respectively, with $\mathbb{E}\{|x_u^c [n]|^2\} = p_u [n]$ and $\mathbb{E}\{|x_u^b [n]|^2\} = 1$. Then, the received signals at the UAV and ship can be expressed, respectively, in the following equations:

$$y_u^c [n] = \check{\mathbf{w}}_u^H [n] \check{\mathbf{h}}_u^c [n] \alpha_u [n] \sqrt{p_u [n]} x_u^c [n] + \check{\mathbf{w}}_u^H [n] \mathbf{n}_c [n] \quad (10)$$

$$y_u^b [n] = (\mathbf{h}_u^b [n])^H \tilde{\mathbf{f}}_u [n] \alpha_u [n] x_u^b [n] + n_b [n] \quad (11)$$

where $\check{\mathbf{w}}_u [n]$ and $\tilde{\mathbf{f}}_u [n]$ are, respectively, the UAV's combiner and precoder vectors for receiving the uplink signal and transmitting the backhaul signal, and the corresponding channels, $\check{\mathbf{h}}_u^c [n]$ and $\mathbf{h}_u^b [n]$, have the similar form with $\check{\mathbf{h}}_u^p [n]$, as shown in (9). The achieved data rates at the n th TS over uplink and backhaul are given, respectively, by

$$R_u^c [n] = \gamma_u^c [n] \log \left(1 + \frac{\alpha_u [n] p_u [n] |\check{\mathbf{w}}_u^H [n] \check{\mathbf{h}}_u^c [n]|^2}{N_0 \|\check{\mathbf{w}}_u [n]\|^2} \right) \quad (12)$$

$$R_u^b [n] = \gamma_u^b [n] \log \left(1 + \frac{\alpha_u [n] |(\mathbf{h}_u^b [n])^H \tilde{\mathbf{f}}_u [n]|^2}{N_0} \right). \quad (13)$$

III. PROBLEM FORMULATION

We deal with a practical scenario, in which the ship reaches a specific sea area and releases the UAV to collect the monitoring data from all the buoys therein. Since buoys used for monitoring in the open sea are energy-limited, the energy for implementing uplink transmissions is no larger than that is harvested during the power transferring stage. During the UAV mission period, the objective is to collect the monitoring data as much as possible. To avoid the data deficiency of any observation area, the amount of collected data from each buoy should be no less than a threshold. In light of the timeliness of monitoring data, the UAV promptly forwards part of the collected data to the central ship during flying.

As the LoS path dominates the channel between any buoy and the UAV as well as only one buoy can be served by the UAV at each TS, the UAV adopts the matched-filter beamforming strategy to transmit/receive signals, i.e.,

$$\begin{cases} \mathbf{f}_u [n] = \mathbf{v}_u [n] = \sqrt{P_{\text{UAV}} \mathbf{a}_{\text{tx}}} (\mathbf{c}_{\text{UAV}} [n], \mathbf{c}_u [n]) \\ \mathbf{w}_u [n] = \check{\mathbf{w}}_u [n] = \mathbf{a}_{\text{rx}} (\mathbf{c}_{\text{UAV}} [n], \mathbf{c}_u [n]) \\ \tilde{\mathbf{f}} [n] = \sqrt{P_{\text{UAV}} \mathbf{a}_{\text{tx}}} (\mathbf{c}_{\text{UAV}} [n], \mathbf{c}_{\text{ship}}) \end{cases} \quad (14)$$

²In practice, the nonlinear RF-to-dc conversion model is more accurate, especially when the input power is extremely low or approaches the saturation region [32]. However, if the system operates within the quasi-linear region, the deviation introduced by the linear model is very limited. Therefore, this work adopts the linear model to simplify the optimization formulation, and the proposed algorithm can also be extended to the nonlinear case.

where P_{UAV} is the UAV's maximum transmit power, and \mathbf{c}_{ship} is the 3-D coordinate of the ship, which remains unchanged during mission period. The optimization problem to achieve the mission objective can be formulated as

$$(P1) : \max_{\{\alpha_u[n], \{\gamma_u^x[n]\}, \{\mathbf{c}_{\text{UAV}}[n]\}, \{p_u[n]\}} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] \quad (15a)$$

$$\text{s.t. } \alpha_u[n] \in \{0, 1\}, \sum_{u \in \mathcal{U}} \alpha_u[n] \leq 1 \quad \forall n \quad (15b)$$

$$0 \leq \gamma_u^x[n] \leq 1, \sum_x \gamma_u^x[n] \leq 1 \quad \forall n \quad (15c)$$

$$x \in \{s, p, c, b\} \quad (15c)$$

$$R_u^s[n] \geq \alpha_u[n] \Gamma_s^{\text{th}} \quad \forall u, n \quad (15d)$$

$$P_u^p[n] \geq \gamma_u^c[n] \alpha_u[n] p_u[n] \quad \forall u, n \quad (15e)$$

$$\frac{1}{N} \sum_{n=1}^N R_u^c[n] \geq \Gamma_c^{\text{th}} \quad \forall u \quad (15f)$$

$$R_u^b[n] \geq \chi_u R_u^c[n] \quad \forall u, n \quad (15g)$$

$$\|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{UAV}}[n-1]\| \leq V_{\text{max}} T_t \quad \forall n > 1 \quad (15h)$$

$$\mathbf{c}_{\text{UAV}}^1 = \mathbf{c}_{\text{UAV}}[1], \mathbf{c}_{\text{UAV}}^F = \mathbf{c}_{\text{UAV}}[N]. \quad (15i)$$

Constraint (15b) specifies that only one buoy can be served by the UAV within each TS. Constraint (15c) states the time allocation among different stages, which may vary at different TSs. To meet the sensing QoS, (15d) states that the sensing MI of buoy u within the n th TS (if $\alpha_u[n] = 1$) must be larger than a minimum estimation threshold Γ_s^{th} . Due to the limited service life of buoys, buoys only use the harvested power to carry out uplink transmission, as stated in (15e). Constraint (15f) states that the collected data from each buoy must be larger than a rate threshold Γ_c^{th} . Constraint (15g) states that χ_u ($0 \leq \chi_u \leq 1$) proportion of the collected data by buoy u within each TS is offloaded to the ship. Constraints (15h) and (15i) specify the UAV's maximum flying distance within one TS and its initial and final positions.

IV. SOLUTIONS TO THE OPTIMIZATION PROBLEM

The proposed optimization (P1), jointly considering the user association between buoys and the UAV, time allocation among different tasks, buoys' transmit power, and UAV's trajectory, is a mixed-integer nonconvex problem. In particular, unlike the terrestrial UAV-assisted ISAC, the existence of sea clutters in the proposed maritime networks further introduces extreme difficulty to (P1).

To this end, we reformulate the expressions of $R_u^s[n]$, $P_u^p[n]$, $R_u^c[n]$, and $R_u^b[n]$ using the beamforming strategy (14). Specifically, given (7) and (14), $R_u^c[n]$ is reformulated as (16), shown

at the bottom of the page, where $G_{\text{gain}}^s = P_{\text{UAV}} G_{\text{UAV}}^{\text{tx}} G_{\text{UAV}}^{\text{rx}}$ and $\bar{d}_u[n] = ((x_{\text{UAV}}[n] - x_u[n])^2 + (y_{\text{UAV}}[n] - y_u[n])^2)^{1/2}$ denotes the 2-D Euclidean distance between the UAV and buoy u . Moreover, under the assumption of moderate sea conditions, given (12), $R_u^c[n]$ can be reformulated as

$$R_u^c[n] = \gamma_u^c[n] \alpha_u[n] \log \left(1 + \frac{p_u[n] G_{\text{gain}}^c \lambda^2}{N_0 4\pi^2 d_u^2[n]} \right) \quad (17)$$

the derivation of which is given in Appendix A, where $G_{\text{gain}}^c = G_{\text{UAV}}^{\text{tx}} G_{\text{UAV}}^{\text{rx}}$. Similarly, given (8), $P_u^p[n]$ can be rewritten as

$$P_u^p[n] = \frac{\alpha_u[n] \gamma_u^p[n] \xi_u G_{\text{gain}}^p \lambda^2}{4\pi^2 d_u^2[n]} \quad (18)$$

where $G_{\text{gain}}^p = P_{\text{UAV}} G_{\text{UAV}}^{\text{tx}} G_{\text{UAV}}^{\text{rx}}$. Also, given (13), by defining $G_{\text{gain}}^b = P_{\text{UAV}} G_{\text{UAV}}^{\text{tx}} G_{\text{ship}}^{\text{rx}}$ and $d_b[n] = \|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{ship}}\|$, $R_u^b[n]$ can be reformulated as

$$R_u^b[n] = \gamma_u^b[n] \alpha_u[n] \log \left(1 + \frac{G_{\text{gain}}^b \lambda^2}{N_0 4\pi^2 d_b^2[n]} \right). \quad (19)$$

Remark 3: Based on the derivation in Appendix A, it is implied in (17) and (18) that, for moderate sea states and adequate UAV flight height, the effect of position deviation of buoys caused by sea waves on uplink communication and wireless charging within one TS is negligible. This will be further verified in Section V-B. Note that, even in severer sea state, the length of TS can be further shortened to guarantee that the fluctuation of buoys within one TS is limited in the range of UAV's half-power beamwidth. However, the position drift of buoys accumulates over TSs, leading to beam misalignment between the UAV and buoys. Hence, the sensing stage at the beginning of each TS is essential for position recalibration.

It can be found that the association indicators $\{\alpha_u[n]\}$ and the time allocation variables $\{\gamma_u^x[n]\}$ coexist inside expressions (16)–(19), and they satisfy the relationship

$$\begin{cases} \gamma_u^s[n] + \gamma_u^p[n] + \gamma_u^c[n] + \gamma_u^b[n] = 1, & \text{if } \alpha_u[n] = 1 \\ \gamma_u^s[n] + \gamma_u^p[n] + \gamma_u^c[n] + \gamma_u^b[n] = 0, & \text{if } \alpha_u[n] = 0. \end{cases} \quad (20)$$

To deal with the tight coupling of $\{\alpha_u[n]\}$ and $\{\gamma_u^x[n]\}$, we divide (P1) into two subproblems, which are given as

$$(P1.1) : \max_{\mathcal{S}_1} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] \quad (21a)$$

$$\text{s.t. (15b), (15d) – (15i)} \quad (21b)$$

where $\mathcal{S}_1 = \{\alpha_u[n], \mathbf{c}_{\text{UAV}}[n], p_u[n]\}$, which is also the set of optimization variables in (P2) of (24), and

$$(P1.2) : \max_{\mathcal{S}_2} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] \quad (22a)$$

$$\text{s.t. (15c) – (15g)} \quad (22b)$$

$$R_u^s[n] = \gamma_u^s[n] \alpha_u[n] \log \left(1 + \frac{G_{\text{gain}}^s \lambda^2 \psi_u R B \bar{d}_u[n]}{0.886 c G_{\text{gain}}^s \lambda^2 \Gamma_s e^{-\Gamma_s \left(\frac{\bar{d}_u[n]}{v_{\text{UAV}} - z_u[n]} \right)^2} d_u^2[n] + N_0 R B 4\pi^3 d_u^4[n] \bar{d}_u[n]} \right) \quad (16)$$

where $\mathcal{S}_2 = \{\gamma_u^x[n]\}$. Given \mathcal{S}_2 , (P1.1) optimizes \mathcal{S}_1 , i.e., the buoy-UAV association and UAV trajectory, while given \mathcal{S}_1 , (P1.2) optimizes \mathcal{S}_2 , i.e., the time allocation within each TS. Using AO, (P1.1) and (P1.2) are iteratively processed to finally achieve the optimal solution of (P1). Obviously, once the set \mathcal{S}_1 is given, the convex optimization problem (P1.2) can be readily addressed. To solve (P1.1), however, its objective function as well as the involved constraints (15d)–(15g) must be transformed into tractable convex ones.

A. Solution to (P1.1)

1) *Optimization Strategy*: The binary variables $\{\alpha_u[n]\}$ hinder the convexification of the objective function and the related constraints in (P1.1). To deal with $\{\alpha_u[n]\}$, it is first converted into a continuous form, i.e., $0 \leq \alpha_u[n] \leq 1$, and the slack continuous variables $\{\bar{\alpha}_u[n]\}$ are introduced to yielding two equivalent equality constraints given by

$$\alpha_u[n](1 - \bar{\alpha}_u[n]) = 0, \quad \alpha_u[n] = \bar{\alpha}_u[n] \quad \forall u, n. \quad (23)$$

Hence, although $\alpha_u[n]$ is relaxed to a continuous form, it can only be 1 or 0 owing to the equality constraints (23).

After integrating the two equality constraints in (23) into the objective function of (P1.1) as penalty terms, a new optimization problem (P2) is formed

$$(P2) : \max_{\mathcal{S}_1, \{\bar{\alpha}_u[n]\}} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] - \frac{1}{\eta} \sum_{n=1}^N \sum_{u=1}^U ((\alpha_u[n](1 - \bar{\alpha}_u[n]))^2 + (\alpha_u[n] - \bar{\alpha}_u[n])^2) \quad (24a)$$

s.t. (15d) – (15i) (24b)

where $\eta > 0$ is a tunable parameter to control the optimization progress. As η is gradually reduced to 0, the optimization result of (P2) is equal to that of (P1.1). The optimization problem (P2) consists of several sets of variables, $\{\alpha_u[n]\}$, $\{\bar{\alpha}_u[n]\}$, $\{p_u[n]\}$ and $\{\mathbf{c}_{\text{UAV}}[n]\}$. Hence, it can be solved by iteratively fixing some of these sets and optimizing the remaining ones.

As $\{\bar{\alpha}_u[n]\}$ only exist in the objective function and rely on the given $\{\alpha_u[n]\}$, we first derive the optimal $\{\bar{\alpha}_u[n]\}$ through differentiation as $\bar{\alpha}_u^{\text{opt}}[n] = (\alpha_u[n] + \alpha_u^2[n]) / (1 + \alpha_u^2[n]) \quad \forall u, n$. Given $\mathcal{S}_3 = \{\alpha_u[n], \bar{\alpha}_u[n]\}$, the optimization of UAV trajectory and buoys' transmit power is formulated as

$$(P2.1) : \max_{\mathcal{S}_4} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] \quad (25a)$$

s.t. (15d) – (15i) (25b)

where $\mathcal{S}_4 = \{\mathbf{c}_{\text{UAV}}[n], p_u[n]\}$. Given \mathcal{S}_4 and $\{\bar{\alpha}_u[n]\}$, the optimization of $\{\alpha_u[n]\}$ can be readily solved, as the objective function is concave and the involved constraints are linear. Obviously, the challenge of solving the optimization problem (P2) resides in dealing with (P2.1), since the objective function and most of constraints of (P2.1) are nonconvex. Therefore, we address the convexification of the objective function and constraints of (P2.1).

2) *Objective Function Convexification for Subproblem (P2.1)*: The objective function of (P2.1) has a form of $\sum_{n=1}^N \sum_{u=1}^U \alpha_u[n] \log(1 + (\mathcal{A}_u[n]) / (\mathcal{B}_u[n]))$, where $\mathcal{A}_u[n] = p_u[n] G_{\text{gain}}^c \lambda^2$ and $\mathcal{B}_u[n] = N_0 4\pi^2 d_u^2[n]$ are both linear and convex functions. According to the principle of QT [34], it is equivalent to $\sum_{n=1}^N \sum_{u=1}^U \alpha_u[n] \log(1 + 2\varpi_u[n] (\mathcal{A}_u[n])^{1/2} - \varpi_u^2[n] \mathcal{B}_u[n])$, which is a concave function, when either \mathcal{S}_4 or $\varpi_u[n] \in \mathbb{R}$ is fixed.

3) *Constraints Convexification for Subproblem (P2.1)*: To deal with the nonconvex constraints (15d) and (15g) with $\alpha_u[n] \neq 0$, we first rewrite them, respectively, as

$$C_1 e^{-\frac{\Gamma_s d_u^2[n]}{\bar{z}}} \left(\frac{\bar{d}_u^2[n] + \bar{z}}{\bar{d}_u[n]} \right) + C_2 (\bar{d}_u^2[n] + \bar{z})^2 \leq C_u[n] \quad (26)$$

$$\left(1 + \frac{C_3 p_u[n]}{d_u^2[n]} \right)^{r_u^{\text{bc}}[n]} - \left(1 + \frac{C_4}{d_b^2[n]} \right) \leq 0 \quad (27)$$

where $\bar{z} = (z_{\text{UAV}} - z_u[n])^2$, $C_u[n] = (G_{\text{gain}}^s \lambda^2 \psi_u RB) / (2(\Gamma_s^{\text{th}} / \gamma_u^s[n]) - 1)$, $C_1 = 0.886c G_{\text{gain}}^s \lambda^2 \Gamma_s$, $C_2 = N_0 RB 4\pi^3$, $C_3 = (G_{\text{gain}}^c \lambda^2) / (N_0 4\pi^2)$, $C_4 = (G_{\text{gain}}^b \lambda^2) / (N_0 4\pi^2)$, and $r_u^{\text{bc}}[n] = (\gamma_u^c[n] \chi_u) / (\gamma_u^b[n])$. Define the left-hand sides of (26) and (27) as $\Lambda_u(\mathbf{c}_{\text{UAV}}[n])$ and $\Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])$, respectively, which can be iteratively approximated using the first-order Taylor expansion (FOTE) at local point $\mathbf{c}_{\text{UAV}}^{(k)}[n]$ and $p_u^{(k)}[n]$, denoted as $\tilde{\Lambda}_u(\mathbf{c}_{\text{UAV}}[n])$ and $\tilde{\Pi}_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])$. More details of the derivation can be found in Appendix B. In addition, the similar expressions of $(\partial \Lambda_u(\mathbf{c}_{\text{UAV}}[n]) / (\partial y_{\text{UAV}}[n]))$ and $(\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]) / (\partial y_{\text{UAV}}[n]))$ are omitted for the sake of conciseness. As long as the values of $\|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{UAV}}^{(k-1)}[n]\|$ and $|p_u[n] - p_u^{(k-1)}[n]|$ are sufficiently small, constraints (15d) and (15g) can be substituted, respectively, by

$$\tilde{\Lambda}_u(\mathbf{c}_{\text{UAV}}[n]) - C_u[n] \leq 0 \quad \forall u, n \quad (28)$$

$$\tilde{\Pi}_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]) \leq 0 \quad \forall u, n. \quad (29)$$

Next, when $\alpha_u[n] \neq 0$, constraint (15e) can be re-expressed as $d_u^2[n] - (J_u[n]) / (p_u[n]) \leq 0$, which has a form of convex-minus-convex, and $J_u[n] = (\gamma_u^p[n] \xi_u G_{\text{gain}}^p \lambda^2) / (\gamma_u^c[n] 4\pi^2)$. We can resort to the inner convex approximation method and obtain the concave lower bound of $(J_u[n]) / (p_u[n])$, which is given by [35]

$$\frac{J_u[n]}{p_u[n]} \geq \frac{2J_u[n]}{p_u^{(k)}[n]} - \frac{p_u[n] J_u[n]}{(p_u^{(k)}[n])^2}. \quad (30)$$

Consequently, constraint (15e) can be iteratively replaced with the following convex one:

$$d_u^2[n] - \frac{2J_u[n]}{p_u^{(k)}[n]} + \frac{p_u[n] J_u[n]}{(p_u^{(k)}[n])^2} \leq 0 \quad \forall u, n. \quad (31)$$

Although constraint (15f) is mainly composed of $R_u^c[n]$, the QT used to convexify the objective function of (P2.1) is inapplicable to it, since the initial iteration value may not satisfy the constraint. In this regard, we still rely on FOTE to approximately transformed the nonconvex expression into a linear form $\tilde{R}_u^c[n]$, as shown in Appendix B. Then, (15f) can be substituted with

$$\Gamma_c^{\text{th}} - \frac{1}{N} \sum_{n=1}^N \tilde{R}_u^c[n] \leq 0 \quad \forall u. \quad (32)$$

Up to this point, all the nonconvex constraints have been transformed into the convex ones, and we can proceed to carry out the optimization implementation.

4) *Implementation of Subproblem (P2.1)*: After the transformation of objective function and constraints, the optimization problem (P2.1) is equivalent to

$$(P2.2) : \max_{\mathcal{S}_4, \{\varpi_u[n]\}} \sum_{n=1}^N \sum_{u=1}^U \alpha_u[n] \log \left(1 + 2\varpi_u[n] \sqrt{\mathcal{A}_u[n]} - \varpi_u^2[n] \mathcal{B}_u[n] \right) \quad (33a)$$

$$\text{s.t. (28), (29), (31), (32), (15h), (15i)} \quad (33b)$$

$$\left| \mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{UAV}}^{(k-1)}[n] \right| \leq \vartheta^{(k)} \nu_c \quad \forall n \quad (33c)$$

$$\left| p_u[n] - p_u^{(k-1)}[n] \right| \leq \vartheta^{(k)} \nu_p \quad \forall u, n \quad (33d)$$

where $\vartheta^{(k)}$ is the adjustable factor to control the optimization step size at the k th iteration, and ν_c and ν_p are the unique scales of $\mathbf{c}_{\text{UAV}}[n]$ and $p_u[n]$, respectively. Since either \mathcal{S}_4 or $\{\varpi_u[n]\}$ is fixed, the objective function is concave, and given \mathcal{S}_4 , the optimal $\varpi_u^{(k)}[n]$ is $\varpi_u^{(k)}[n] = ((\mathcal{A}_u^{(k-1)}[n])^{1/2})/(\mathcal{B}_u^{(k-1)}[n]) \forall n, u$. Therefore, the objective function can be iteratively updated during the optimization process.

However, to start the optimization procedure, a feasible initial point ($\mathbf{c}_{\text{UAV}}^{(0)}[n], p_u^{(0)}[n]$) is required. In particular, it is impractical to randomly initialize $p_u^{(0)}[n]$, as each buoy's transmit power, i.e., $p_u^{(0)}[n]$, is harvested during UAV wireless charging, and the amount of which heavily depends on the UAV trajectory. Hence, the initialization should be done in the way that satisfies the constraints in (33b). Thus, a feasible initial point can be obtained by solving the following optimization:

$$(P2.2) : \min_{\boldsymbol{\omega}} \sum_{u=1}^U \left(\sum_{n=1}^N (\omega_{1,u}[n] + \omega_{2,u}[n] + \omega_{3,u}[n]) + \omega_{4,u} \right) \quad (34a)$$

$$\text{s.t. (15h)(15i)} \quad (34b)$$

$$\tilde{\Lambda}_u(\mathbf{c}_{\text{UAV}}[n]) - C_u[n] \leq \omega_{1,u}[n] \quad \forall u, n \quad (34c)$$

$$\tilde{\Pi}_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]) \leq \omega_{2,u}[n] \quad \forall u, n \quad (34d)$$

$$d_u^2[n] - \frac{2J_u[n]}{p_u^{(k-1)}[n]} + \frac{p_u[n]J_u[n]}{(p_u^{(k-1)}[n])^2} \leq \omega_{3,u}[n] \quad \forall u, n \quad (34e)$$

$$\Gamma_c^{\text{th}} - \frac{1}{N} \sum_{n=1}^N \tilde{R}_u^c[n] \leq \omega_{4,u} \quad \forall u \quad (34f)$$

$$\omega_{i,u}[n] \geq 0, 1 \leq i \leq 3, \omega_{4,u} \geq 0 \quad \forall u, n \quad (34g)$$

where $\boldsymbol{\omega} = \{\{\omega_{1,u}[n], \omega_{2,u}[n], \omega_{3,u}[n]\}, \omega_{4,u}\}$.

Here, we should mention that (P2.2) is still an iterative optimization problem and needs start-up points to commence. With the aid of slack variables $\boldsymbol{\omega}$, the start-up points of (P2.2) can be randomly selected,³ and the reasonable ones can make (P2.2) quickly converge. Then, it can be seen that the feasible

³If buoys are deployed in a regular manner, the start-up points of UAV trajectory can be defined along the buoy distribution.

initial point ($\mathbf{c}_{\text{UAV}}^{(0)}[n], p_u^{(0)}[n]$) is found when the value of (34a) is close to zero.

B. Algorithmic Analysis for Solving (P1)

The optimization process of solving (P1) is presented in Algorithm 1, which is based on AO and is composed of outer- and inner-layer iterations. Let i and j be the indexes of the outer- and inner-layer iterations, respectively, and denote $\mathcal{R} = \sum_{n=1}^N \sum_{u=1}^U R_u^c[n]$. Within each inner-layer iteration, the alternating update among $\{\alpha_u[n]\}, \{\bar{\alpha}_u[n]\}$ and $\{p_u[n], \mathbf{c}_{\text{UAV}}[n]\}$ leads to an increased rate, i.e., $\mathcal{R}^{(j_1)} \leq \mathcal{R}^{(j_2)}$, $j_1 \leq j_2$. After the inner-layer iteration, the update of time allocation further enhances the rate performance, leading to $\mathcal{R}^{(i_1)} \leq \mathcal{R}^{(i_2)} \leq \dots \leq \mathcal{R}^{(j_{\text{end}})} \leq \mathcal{R}^{(i_2)}$, $i_1 \leq i_2$. The convergence behavior of the adopted QT and inner convex approximation has been demonstrated in [35] and [36], and hence, the convergence of Algorithm 1 is guaranteed.

Algorithm 1 Iterative Solution to Optimization (P1)

```

1 Initialization:
2 Give buoys' positions  $\mathbf{c}_u[n], \forall u, n$ ;
3 Initialize  $\gamma_u[n], \alpha_u[n], \forall u, n$ ;
4 [Outer Layer Iteration]
5   repeat
6     For current  $\mathcal{S}_2$ , obtain optimal  $\mathcal{S}_1$  by solving
       (P1.1) within following inner layer iteration;
7     [Inner Layer Iteration]
8       repeat
9         Compute  $\bar{\alpha}_u[n] = \frac{\alpha_u[n] + \alpha_u^2[n]}{1 + \alpha_u^2[n]}, \forall u, n$ ;
10        Find initial feasible iteration values
           $\mathbf{c}_{\text{UAV}}^{(0)}[n]$  and  $p_u^{(0)}[n]$  by solving
          (P2.2), and obtain optimal  $\mathbf{c}_{\text{UAV}}[n]$ 
          and  $p_u[n]$  through (P2.2);
11        Derive optimal  $\alpha_u[n]$  when fixing
           $\mathcal{S}_4$  and  $\bar{\alpha}_u[n]$  in (P2);
12        until Objective value of (P2) converges;
13      end
14      For current  $\mathcal{S}_1$ , obtain optimal  $\mathcal{S}_2$  by solving
        (P1.2);
15    until Objective value of (P1) converges;
16 end
Output:  $\{\gamma_u^*[n], \alpha_u^*[n], \mathbf{c}_{\text{UAV}}^*[n], p_u^*[n]\}$ .

```

Remark 4: To deal with (P2.2), the FOTE can be adopted to convexify the nonconvex constraints. However, when the trust region, i.e., $|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{UAV}}^{(k-1)}[n]|$ and $|p_u[n] - p_u^{(k-1)}[n]|$, is large, a problem may arise that even though the transformed constraints, e.g., (28), (29), (31), and (32), are satisfied and (P2.2) is successively solved, the corresponding original constraints, e.g., (15d)–(15g), are violated. We introduce an augmented FOTE (A-FOTE) to address this issue. Specifically, the adjustable factor $\vartheta^{(k)}$ is initialized with a sufficiently big value to enlarge the trust region at the beginning of the iteration. Following each execution of (P2.2), $\vartheta^{(k)}$ is adaptively decreased for the subsequent iteration if either: 1) no feasible solution exists or 2) a feasible solution is found but violates

TABLE II
DEFAULT SIMULATION SYSTEM PARAMETERS

Parameters	Values
UAV power budget (P_{UAV})	36 dBm
Antenna gains of ship, UAV and buoys	(30, 26, 20) dBi
Noise power spectrum density (N_0)	-107 dBm
Energy conversion efficiency ($\xi_u, \forall u$)	0.8
Sea State (κ_s)	3
UAV flight height (z_{UAV})	10 m
Central frequency and bandwidth	5 GHz, 10 MHz
Minimum communication QoS (Γ_c^{th})	1 bps/Hz
Ratio of data offloaded to the ship ($\chi_u, \forall u$)	0.2
Minimum sensing QoS (Γ_s^{th})	1 bps/Hz

any of the original constraints. Using this A-FOTE, the inner-layer iteration is guaranteed to converge in a trusted way.

According to [37], the computational complexity of each inner-layer iteration is given by $L_1 = \mathcal{O}((6N + U)^{2.5}(2N)^2 + (3N + U)^4)$, and the computational complexity of optimizing (P1.2) is given by $L_2 = \mathcal{O}((4N + U)^{2.5}(4N)^2)$. Hence, the total computational complexity of Algorithm 1 is $\rho_o(\rho_i L_1 + L_2)$, where ρ_i and ρ_o denote the numbers of inner- and outer-layer iterations, respectively.

V. SIMULATION RESULTS

We evaluate the performance of the UAV-enabled ISCPB in maritime monitoring networks and demonstrate the effectiveness of the proposed joint optimization method for UAV trajectory, time allocation, user association, and power scheduling. In the simulation, we consider a practical buoy-based monitoring scenario in open sea, where ten buoys are uniformly distributed along a circular arc with a radius of 500 m and centered at the coordinate of (500, 0, 0) m to form a buoy monitoring array. The ship that establishes the backhaul link with the UAV situates at (500, -20, 0) m. The UAV equipped with an 8×8 UPA antenna array takes off from the coordinate of (0, -10, 10) m and flies toward the destination point of (1000, -10, 10) m with a maximum speed of 30 m/s. The whole mission period is discretized into 100 TSs, each of which contributes 1 s. Unless otherwise specified, the default system's parameters listed in Table II are used.

A. Performance of Proposed Optimization Method

To demonstrate the effectiveness of our proposed method for jointly optimizing the UAV trajectory, time allocation, user association, and power scheduling, we evaluate its performance in terms of the achievable rate and sensing MI as well as the convergence behavior. First, we consider the optimization problem (P2.2) and obtain a feasibly initial UAV trajectory, as shown in Fig. 4, that meets the basic QoSs of communication rates for both uplink and backhaul link as well as sensing. It can be observed that as the UAV is only connected with one buoy within each TS, its initial flight trajectory is generally along the circular arc following the buoys' coordinate distribution, such that the link between the UAV and the buoy being served is as short as possible.

Then, we examine the optimization process presented in Algorithm 1. Specifically, the outer-layer iteration process of UAV trajectory is depicted in Fig. 5, which shows that three

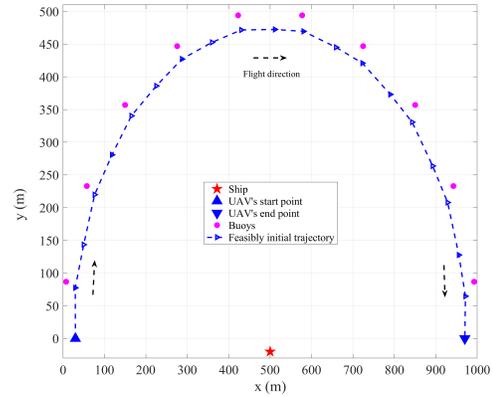


Fig. 4. Diagram of the feasibly initial UAV trajectory.

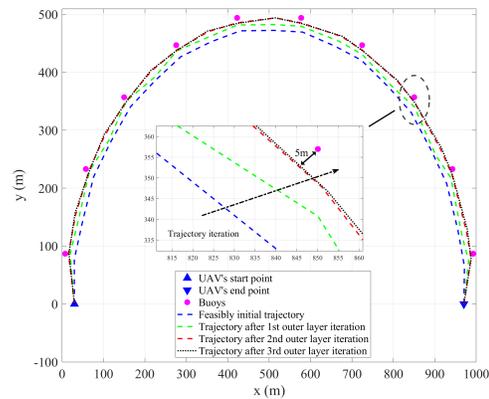


Fig. 5. Outer-layer iteration process of UAV trajectory.

outer-layer iterations are needed to achieve convergence, and as the number of iterations increases, the UAV flies closer to buoys. However, although the shorter distance usually leads to better performance of sensing and communication, the UAV avoids to fly over buoys (FoB). This is because the larger the grazing angle is, the stronger the sea clutter interference becomes, and to decrease the effect of sea clutter, the UAV keeps a feasible horizontal angle with buoys. Also, it can be seen from Fig. 5 that after the third iteration, the shortest distance between the UAV and buoy is slightly over 5 m leading to around 60° grazing angle. This exactly matches the results in Fig. 3, which show that when $\kappa_s = 3$, once the grazing angle is larger than 60° , the effect of sea clutter becomes more severe.

The performance of Algorithm 1 in terms of communication rate, user association, sensing MI, and harvested power is presented in Fig. 6. As shown in Fig. 6 (a), the average rate per TS increases progressively with each outer-layer iteration, ultimately converging at the fourth iteration and attaining over 22 bps/Hz. In addition, compared with the inner-layer iteration (i.e., steps 7–13 in Algorithm 1), the outer-layer iteration adjusts time allocation (i.e., step 14 in Algorithm 1) to bring a more significant improvement to the rate performance. Fig. 6(b) illustrates the dynamic user association among the UAV and buoys. In particular, as the

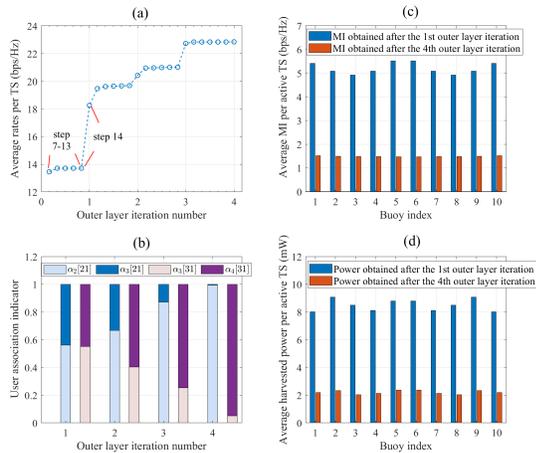


Fig. 6. Optimization performance in terms of (a) communication rate, (b) user association, (c) sensing MI, and (d) harvested power.

optimization process progresses, the UAV is prone to serve buoys 2 and 4 at the 21st and 31st TSs, respectively. Moreover, Fig. 6(c) and (d) shows the achievable performance of MI and harvested power respectively after the 1st and 4th outer-layer iterations, where it can be observed that since the optimization objective is to maximize the uplink communication rates, the average MI/amount of harvested power per active TS⁴ deteriorates to some extent. But, owing to the constraints of (15d) and (15e), the average MI per active TS of each buoy is still larger than the predefined threshold of 1 bps/Hz, while the harvested power of around 2 mW per active TS is sufficient to support the uplink communications.

In addition, as our proposed optimization method relies on the integration of QT and A-FOTE, to demonstrate their effectiveness, we provide a performance comparison against the following two representative benchmarks.

- 1) *Dinkelbach's Transform (DT)* [38]: The objective function of (P2.1) is reformed into $\sum_{n=1}^N \sum_{u=1}^U \alpha_u[n] \log(1 + \mathcal{A}_u[n] - \kappa_u[n] \mathcal{B}_u[n])$ with $\kappa_u[n]$ being iteratively updated to $(\mathcal{A}_u[n]) / (\mathcal{B}_u[n])$ based on DT principle and the A-FOTE is applied.
- 2) *Fixed Optimization Step size (FOS)* [16]: Instead of applying A-FOTE, FOS adopts the fixed optimization stepsize, i.e., $\vartheta^{(k)}$ in (33) is nontunable, while the QT method is used.

As shown in Fig. 7, the proposed approach outperforms the DT method, since DT is specialized in handling single-ratio concave-convex FP problem [34], whereas the objective function in (25) is a multiple-ratio one. Additionally, without the aid of A-FOTE, FOS lags behind the proposed and the DT methods and causes violated constraints [Note that the constraints violation ratio is calculated by (Number of violated constraints)/(Number of all constraints)]. The reason behind is that with the fixed stepsize, the optimization process only explores the feasible points in a limited size of range and fails to converge to an optimum point. Also, if the stepsize is larger than the size of feasible

⁴During the mission period, each buoy is only served by the UAV in certain TSs, and these TSs are denoted as “active TSs.”

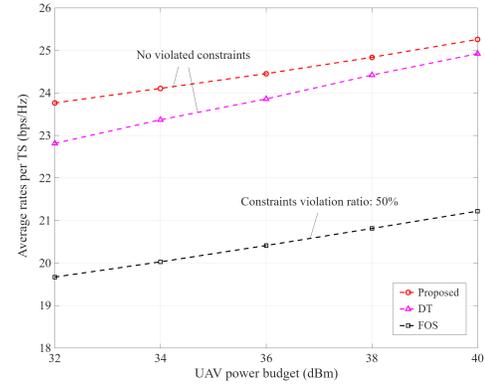


Fig. 7. Effectiveness of the proposed optimization method incorporating QT and A-FOTE compared with benchmarks.

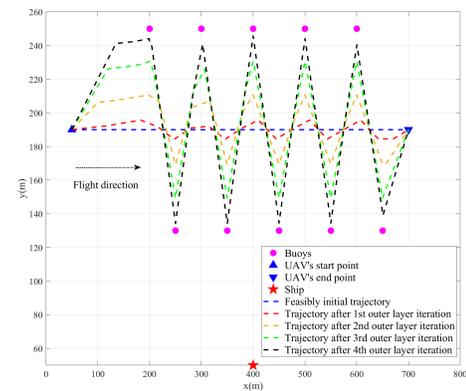


Fig. 8. Convergence behavior of the proposed optimization method for another buoy monitoring array.

region, the optimization is more likely to get stuck in an infeasible region, resulting in the constraints not being satisfied.

B. Impacts of System Structure and States

To demonstrate the robustness of the proposed optimization method, we construct another buoy monitoring array, as shown in Fig. 8, where ten buoys constitute two parallel rows of linear monitoring array. As shown in Fig. 8, with the aid of the proposed method, the optimal UAV trajectory can be obtained after four outer-layer iterations.

Next, we analyze the effect of sea clutter on the performance of the proposed network. As shown in Fig. 9, the UAV passes almost directly over the buoy under sea state 1, since sea clutter only comes from the grazing angle of around 80° – 90° , which is depicted in Fig. 3. When the sea state changes to 2, to avoid the sea clutter reflected from the grazing angle of around 60° – 90° , the UAV moves away from the buoy. When the sea state evolves from 2 to 4, the UAV trajectory does not change too much compared with that of sea state 2. This is because with the increase of sea state, although the sea clutter arises from wider range of grazing angle, its power reduces a lot due to the irregular diffuse reflection caused by sea waves. Since the UAV progressively enlarges its stand-off distance from the buoy when the sea state increases from 1 to 4, the

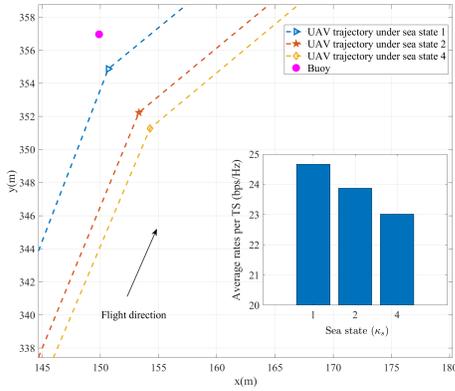


Fig. 9. Effect of sea clutter on the UAV trajectory and rate performance, where the sea states 1, 2, and 4 are considered.

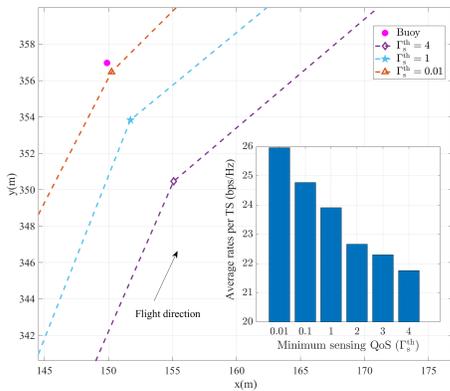


Fig. 10. Effect of minimum sensing QoS on the UAV trajectory and rate performance, where $\Gamma_s^{\text{th}} = 0.01, 0.1, 1, 2, 3, 4$ and $\kappa_s = 3$ are considered.

average rate per TS correspondingly reduces by 1.6 bps/Hz as can be seen from the subfigure of Fig. 9. It can then be inferred that as the sea state becomes worse, the effect of sea clutter may not cause too much rate reduction.

We then investigate the effect of the minimum sensing QoS on the performance of the proposed network. As expected, with the increase of Γ_s^{th} , the UAV allocates more time resource to satisfy the stringent requirement of sensing, leading to the average rate per TS reduction by 4.2 bps/Hz from $\Gamma_s^{\text{th}} = 0.01$ to $\Gamma_s^{\text{th}} = 4$, which is shown in the subfigure of Fig. 10. It can also be seen from in Fig. 10 that increasing minimum sensing QoS causes the UAV moving away from the buoy because of the existing of sea clutter. In particular, to guarantee that all the buoys can be accurately sensed, the UAV keeps a safe distance from buoys so as to avoid being affected by sea clutter during receiving the desired sensing echo.

Recall that in Remark 3, we stated under the assumption of moderate sea states, the effect of buoy's position deviation on the system performance is very limited. To verify it, we conduct the simulation in Fig. 11. Note that "no position deviation" means the buoy's displacement is ignored and the communication rates and harvested power are calculated by (17) and (18). By contrast, (8) and (12) are leveraged to obtain the average performance of communication rates and harvested power, when the effect of displacement is

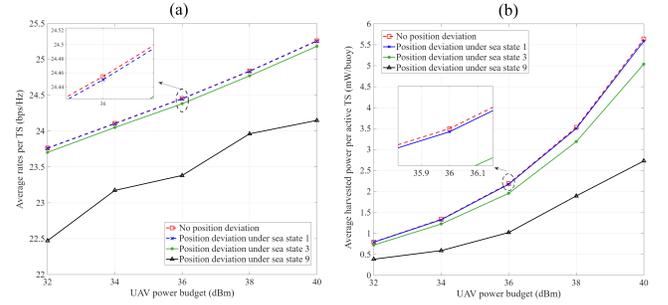


Fig. 11. Effect of buoy's position deviation on the system performance, i.e., (a) communication rates and (b) harvested power, where the statistical wave height and period of different sea states are given as follows: sea state 1 has $H = 0.1$ and $T = 2$; sea state 3 has $H = 0.8$ and $T = 4$; and sea state 9 has $H = 14$ and $T = 20$ [39].

considered. Then, as seen in Fig. 11, when the sea condition is moderate (e.g., the sea state is 1 or 3), the buoy's position deviation has very small effect on system performance, which is consistent with the analysis in Remark 3. In more detail, when the sea state is 3, the communication rates just experience a decrease of 0.1 bps/Hz, and the harvested energy decreases by 0.2 mW per buoy. However, as the sea state increases to 9 (this is an extremely rare condition), the buoy motion becomes very intense, resulting in significant performance degradation.

C. Performance Comparison With Benchmarks

Our scheme is compared with the following benchmarks for UAV-assisted maritime monitoring networks.

- 1) *Fly Over Buoys*: The UAV flies along the arc-shaped buoys monitoring array.
- 2) *Fly-Hover-Fly* [40]: The UAV sequentially reaches the ten optimal hovering points, which are obtained by iteratively solving the transformed optimization problem (P2.1) based on (P2.1)

$$(P2.1) : \max_{\underline{S}_4} \sum_{n=1}^N \sum_{u=1}^U R_u^c[n] \quad (35a)$$

$$\text{s.t. (15d) - (15g), (15i)} \quad (35b)$$

$$\sum_{u=2}^N \|\mathbf{c}_{\text{UAV}}[n] - \mathbf{c}_{\text{UAV}}[n-1]\| \leq V_{\max} T \quad (35c)$$

where $\underline{S}_4 = \{\mathbf{c}_{\text{UAV}}[\mathcal{N}], p_u[\mathcal{N}]\}$ with $\mathcal{N} = \{n_1, \dots, n_{10}\}$ is the set of ten hovering points and power scheduling factors.

- 3) *Triangle Fly*: The UAV flies along a predefined triangle path.
- 4) *Trapezoidal Flight*: The UAV flies along a predefined trapezoidal path.
- 5) *Straight Fly*: The UAV directly flies from the initial position to the destination.
- 6) *Fly With Equal Time Allocation*: The proposed UAV optimization strategy is adopted, while the equal time allocation stays unchanged during mission period.

Unless otherwise stated, the optimization of the other related variables in each benchmark follows the way implemented in Algorithm 1. The optimal UAV trajectory of each scheme

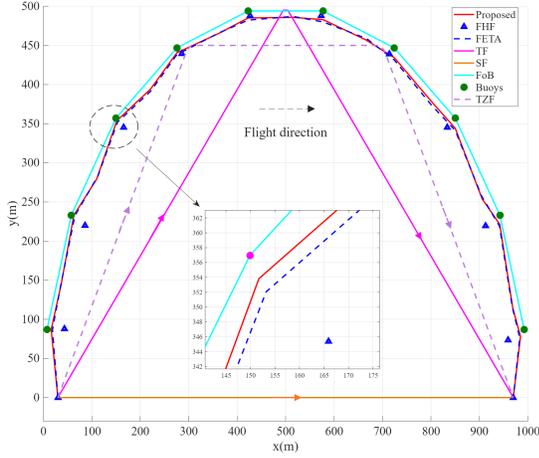


Fig. 12. UAV trajectory comparison between the proposed scheme and six benchmarks.

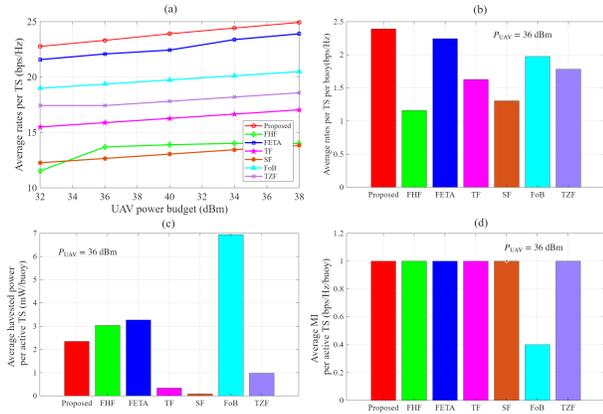


Fig. 13. Comparison between the proposed scheme and six benchmarks, in terms of (a) average rate per TS, (b) average rate per active TS per buoy, (c) average harvested power per active TS per buoy, and (d) average MI per active TS per buoy. The results in (b)–(d) are obtained with $P_{UAV} = 36$ dBm.

is depicted in Fig. 12, and its corresponding performance is provided in Fig. 13. As shown in Fig. 13(a), our scheme outperforms all the benchmarks in terms of average rate per TS during mission period and is able to achieve 25 bps/Hz with 40-dBm UAV's transmit power. Fly with equal time allocation (FETA) without optimizing time allocation lags behind the proposed scheme but is still better than the other benchmarks, as it jointly considers the UAV trajectory, user association, and power scheduling. In triangle fly (TF), trapezoidal flight (TZF), and straight fly (SF), the predefined UAV paths diverge too much from the positions of buoys, leading to lower rates. As seen in Figs. 12 and 13(b), although the optimal hovering points of the UAV are close to buoys, fly-hover-fly (FHF) fails to optimize the path among any of two hovering points, leading to significant decrease of buoys' communication rates.

In terms of charging and sensing performance, FoB is capable of fulfilling the largest charging power, as shown in Fig. 13(c). This is because the UAV flies along the distribution of buoys and therefore has the shortest distance between buoys. On contrary, the UAV flying paths in TF, FZF, and SF are far from buoys causing the unsatisfied performance of wireless charging. However, owing to the effect of sea clutter, FoB,

where the UAV flies too close to buoys, cannot meet the requirement of minimum sensing MI of 1 bps/Hz and only attains 0.4 bps/Hz, while the other schemes are able to satisfy the sensing QoS.

VI. CONCLUSION

To improve the data collection in energy-limited buoy monitoring networks in open sea, we have proposed an ISCPB scheme with specifically designed TDD-like frame structure, in which the UAV first localizes buoys using sensing signal and then carries out wireless charging, data collection, and backhaul transmission. Due to the effect of sea clutters, these multiple transmissions are tightly coupled in terms of time distribution, UAV trajectory, UAV-buoy association, and power scheduling, which has motivated us to design a joint optimization method based on AO, QT, and A-FOTE. Simulation results have demonstrated the effectiveness and good convergence behavior of the joint optimization method. In particular, our ISCPB scheme is capable of achieving an average data rate over 22 bps/Hz per TS relying only on the harvested power, which holds great potential to serve as a paramount data collection solution in open sea monitoring scenarios.

Additionally, in this work, the moderate sea condition is assumed, and thus, the buoy's displacement caused by sea waves is considered negligible. To further enhance the applicability of the proposed ISCPB scheme under harsh maritime environments, future research will focus on developing a robust beam-tracking mechanism to overcome the impact of ocean wave fluctuations and ensure reliable communication and powering performance.

APPENDIX A DERIVATION OF $R_u^c[n]$

Substituting $\tilde{\mathbf{w}}_u[n]$ by $\mathbf{a}_{rx}(\mathbf{c}_{UAV}[n], \mathbf{c}_u[n])$ yields

$$\begin{aligned}
 R_u^c[n] &= \gamma_u^c[n] \log \left(1 + \left(\frac{\alpha_u[n] p_u[n] G_{\text{gain}}^c \lambda^2}{N_0 4\pi^2 \tilde{d}_u^2[n]} \right) \times \left| \mathbf{a}_{rx}^H(\mathbf{c}_{UAV}[n], \mathbf{c}_u[n]) \mathbf{a}_{rx}(\mathbf{c}_{UAV}[n], \tilde{\mathbf{c}}_u[n]) \right|^2 \right) \\
 &\stackrel{(a)}{=} \gamma_u^c[n] \log \left(1 + \left(\frac{\alpha_u[n] p_u[n] G_{\text{gain}}^c \lambda^2}{N_0 4\pi^2 d_u^2[n] R^4} \right) \times \left| \sum_{r=0}^{R-1} e^{-j2\pi r \frac{e_x[n]}{2d_u^2[n]}} \sum_{r=0}^{R-1} e^{-j2\pi r \frac{e_y[n]}{2d_u^2[n]}} \right|^2 \right) \\
 &\stackrel{(b)}{=} \gamma_u^c[n] \log \left(1 + \left(\frac{\alpha_u[n] p_u[n] G_{\text{gain}}^c \lambda^2}{N_0 4\pi^2 d_u^2[n] R^4} \right) \times \left(\frac{\sin(R\hat{d}_u[n] e_x[n]) \sin(R\hat{d}_u[n] e_y[n])}{\sin(\hat{d}_u[n] e_x[n]) \sin(\hat{d}_u[n] e_y[n])} \right)^2 \right) \\
 &\stackrel{(c)}{=} \gamma_u^c[n] \log \left(1 + \left(\frac{\alpha_u[n] p_u[n] G_{\text{gain}}^c \lambda^2}{N_0 4\pi^2 d_u^2[n]} \right) \right) \quad (36)
 \end{aligned}$$

$$\begin{aligned} \Lambda_u(\mathbf{c}_{\text{UAV}}[n]) &\approx \tilde{\Lambda}_u(\mathbf{c}_{\text{UAV}}[n]) \triangleq \Lambda_u(\mathbf{c}_{\text{UAV}}^{(k-1)}[n]) + \left. \frac{\partial \Lambda_u(\mathbf{c}_{\text{UAV}}[n])}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n]} (x_{\text{UAV}}[n] - x_{\text{UAV}}^{(k-1)}[n]) \\ &\quad + \left. \frac{\partial \Lambda_u(\mathbf{c}_{\text{UAV}}[n])}{\partial y_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n]} (y_{\text{UAV}}[n] - y_{\text{UAV}}^{(k-1)}[n]) \end{aligned} \quad (37)$$

$$\begin{aligned} \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]) &\approx \tilde{\Pi}_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]) \triangleq \Pi_u(\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]) + \left. \frac{\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} \\ &\quad \times (x_{\text{UAV}}[n] - x_{\text{UAV}}^{(k-1)}[n]) + \left. \frac{\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])}{\partial y_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} (y_{\text{UAV}}[n] - y_{\text{UAV}}^{(k-1)}[n]) \\ &\quad + \left. \frac{\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])}{\partial p_u[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} (p_u[n] - p_u^{(k-1)}[n]) \end{aligned} \quad (38)$$

$$\begin{cases} \left. \frac{\partial \Lambda_u(\mathbf{c}_{\text{UAV}}[n])}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n]} = \left(C_1 e^{-\frac{\Gamma_s \bar{d}_u^2[n]}{\bar{z}}} \left(-\frac{2\Gamma_s(\bar{d}_u^2[n] + \bar{z})}{\bar{z}} + \frac{\bar{d}_u^2[n] - \bar{z}}{\bar{d}_u^2[n]} \right) + 4C_2 \bar{d}_u[n] (\bar{d}_u^2[n] + \bar{z}) \right) \left(\frac{x_{\text{UAV}}[n] - x_u[n]}{d_u[n]} \right) \\ \left. \frac{\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} = -2r_u^{\text{bc}}[n] \left(1 + \frac{C_3 p_u[n]}{d_u^2[n]} \right)^{r_u^{\text{bc}}[n]-1} \frac{C_3 p_u[n] (x_{\text{UAV}}[n] - x_u[n])}{d_u^2[n]} - \frac{2C_4}{d_u^2[n]} (x_{\text{UAV}}[n] - x_b[n]) \\ \left. \frac{\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])}{\partial p_u[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} = r_u^{\text{bc}}[n] \left(1 + \frac{C_3 p_u[n]}{d_u^2[n]} \right)^{r_u^{\text{bc}}[n]-1} \frac{C_3}{d_u^2[n]} \end{cases} \quad (39)$$

$$\begin{aligned} R_u^c[n] &\approx \tilde{R}_u^c[n] \triangleq R_u^c(\mathbf{c}_{\text{UAV}}^{(k-1)}[n]) + \left. \frac{\partial R_u^c[n]}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} (x_{\text{UAV}}[n] - x_{\text{UAV}}^{(k-1)}[n]) \\ &\quad + \left. \frac{\partial R_u^c[n]}{\partial y_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} (y_{\text{UAV}}[n] - y_{\text{UAV}}^{(k-1)}[n]) + \left. \frac{\partial R_u^c[n]}{\partial p_u[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} (p_u[n] - p_u^{(k-1)}[n]) \end{aligned} \quad (40)$$

where (a) is obtained due to the fact that under the moderate sea conditions, the small wave-induced displacement (i.e., $e_x[n]$, $e_y[n]$ and $e_z[n]$) is much shorter than the distance between buoy u and the UAV, i.e., $e_x[n]$, $e_y[n]$, $e_z[n] \ll d_u[n]$, and hence $d_u[n] \approx \bar{d}_u[n]$. To further illustrate this point, we can take sea states 1 and 3 for examples based on the analysis in Remark 2. In particular, according to [39], when the sea state is 1 and 3, the statistical wave height and period can take the values of $\{H = 0.1, T = 2\}$ and $\{H = 0.8, T = 4\}$. The corresponding values of ϵ^2 are obtained as 0.01 and 0.20. Additionally, in practical maritime operations, to avoid the effects of surface waves and sea spray, UAVs typically fly at altitudes of several tens of meters. Therefore, in the cases of moderate sea states, buoy's position deviation is negligible compared with the UAV's flight altitude. Step (b) is obtained using $\sum_{n=0}^{N-1} e^{-j2\pi n x} = (\sin(\pi N x))/(\sin(\pi x))e^{-j\pi x(N-1)}$, where $\hat{d}_u[n] = (\pi)/(2d_u[n])$, and step (c) is derived using $\lim_{x \rightarrow 0} (\sin ax)/x = a$.

and $(\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]))/(\partial p_u[n])$ are given in (39), as shown at the top of the page.

B. Transformed Expression of $R_u^c[n]$

Based on FOTE, $R_u^c[n]$ (40), as shown at the top of the page, can be approximately transformed into $\tilde{R}_u^c[n]$, shown in (40), where

$$\begin{cases} \left. \frac{\partial R_u^c[n]}{\partial x_{\text{UAV}}[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} = \frac{-2\gamma_u^c[n] \alpha_u[n] p_u[n] G_{\text{gain}}^c \lambda^2 (x_{\text{UAV}}[n] - x_u[n])}{\ln^2(N_0 4\pi^2 d_u^4[n] + p_u[n] G_{\text{gain}}^c \lambda^2 d_u^2[n])} \\ \left. \frac{\partial R_u^c[n]}{\partial p_u[n]} \right|_{\mathbf{c}_{\text{UAV}}^{(k-1)}[n], p_u^{(k-1)}[n]} = \frac{\gamma_u^c[n] \alpha_u[n] G_{\text{gain}}^c \lambda^2}{\ln^2(N_0 4\pi^2 d_u^2[n] + p_u[n] G_{\text{gain}}^c \lambda^2)} \end{cases} \quad (41)$$

while the similar form of $(\partial R_u^c[n])/(\partial y_{\text{UAV}}[n])$ is omitted.

APPENDIX B

TRANSFORMED EXPRESSIONS BASED ON FOTE

A. Transformed Expression of $\Lambda_u(\mathbf{c}_{\text{UAV}}[n])$

According to the principle of FOTE, $\Lambda_u(\mathbf{c}_{\text{UAV}}[n])$ and $\Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])$ can be approximately transformed into $\tilde{\Lambda}_u(\mathbf{c}_{\text{UAV}}[n])$ and $\tilde{\Pi}_u(\mathbf{c}_{\text{UAV}}[n], p_u[n])$, shown in (37) and (38), as shown at the top of the page, respectively, where $(\partial \Lambda_u(\mathbf{c}_{\text{UAV}}[n]))/(\partial x_{\text{UAV}}[n])$, $(\partial \Pi_u(\mathbf{c}_{\text{UAV}}[n], p_u[n]))/(\partial x_{\text{UAV}}[n])$

REFERENCES

- [1] Y. Liu, *Coastal Ocean Observing Systems*. New York, NY, USA: Academic, 2015.
- [2] P. Wang, X. Tian, T. Peng, and Y. Luo, "A review of the state-of-the-art developments in the field monitoring of offshore structures," *Ocean Eng.*, vol. 147, pp. 148–164, Jan. 2018.

- [3] M. Lin and C. Yang, "Ocean observation technologies: A review," *Chin. J. Mech. Eng.*, vol. 33, no. 1, pp. 1–18, Dec. 2020.
- [4] X. André, B. Moreau, and S. Le Reste, "Argos-3 satellite communication system: Implementation on the arvor oceanographic profiling floats," *J. Atmos. Ocean. Technol.*, vol. 32, no. 10, pp. 1902–1914, Oct. 2015.
- [5] N. Nomikos, P. K. Gkonis, P. S. Bithas, and P. Trakadas, "A survey on UAV-aided maritime communications: Deployment considerations, applications, and future challenges," *IEEE Open J. Commun. Soc.*, vol. 4, pp. 56–78, 2023.
- [6] R. Ma, R. Wang, G. Liu, W. Meng, and X. Liu, "UAV-aided cooperative data collection scheme for ocean monitoring networks," *IEEE Internet Things J.*, vol. 8, no. 17, pp. 13222–13236, Sep. 2021.
- [7] X. Fu, M. Ren, and X. Liu, "Collaborative data collection in UAV-enabled maritime IoT with constrained aerial base stations," *IEEE Sensors J.*, vol. 25, no. 1, pp. 1593–1610, Jan. 2025.
- [8] Z. Liu, X. Meng, Y. Yang, K. Ma, and X. Guan, "Energy-efficient UAV-aided ocean monitoring networks: Joint resource allocation and trajectory design," *IEEE Internet Things J.*, vol. 9, no. 18, pp. 17871–17884, Sep. 2022.
- [9] N. Nomikos, A. Giannopoulos, A. Kalafatis, V. Özduran, P. Trakadas, and G. K. Karagiannidis, "Improving connectivity in 6G maritime communication networks with UAV swarms," *IEEE Access*, vol. 12, pp. 18739–18751, 2024.
- [10] L. P. Qian, H. Zhang, Q. Wang, Y. Wu, and B. Lin, "Joint multi-domain resource allocation and trajectory optimization in UAV-assisted maritime IoT networks," *IEEE Internet Things J.*, vol. 10, no. 1, pp. 539–552, Jan. 2023.
- [11] X. Li, W. Feng, Y. Chen, C.-X. Wang, and N. Ge, "Maritime coverage enhancement using UAVs coordinated with hybrid satellite-terrestrial networks," *IEEE Trans. Commun.*, vol. 68, no. 4, pp. 2355–2369, Apr. 2020.
- [12] J. Li, T. Hui, W. Shao, D. Yang, and X. Luan, "Performance analysis of satellite communication transmission links in different sea conditions," in *Proc. Int. Conf. Microw. Millim. Wave Technol. (ICMMT)*, May 2024, pp. 1–3.
- [13] F. Abushakra et al., "A miniaturized ultra-wideband radar for UAV remote sensing applications," *IEEE Microw. Wireless Compon. Lett.*, vol. 32, no. 3, pp. 198–201, Mar. 2022.
- [14] J. Palacios, D. De Donno, and J. Widmer, "Tracking mm-wave channel dynamics: Fast beam training strategies under mobility," in *Proc. IEEE INFOCOM-IEEE Conf. Comput. Commun.*, May 2017, pp. 1–9.
- [15] J. Chen, J. Wang, and L. Bai, "DPCS-SDMA: An ISAC-aided MAC protocol for flying ad hoc networks," *IEEE Trans. Veh. Technol.*, vol. 74, no. 8, pp. 13308–13313, Aug. 2025.
- [16] X. Jing, F. Liu, C. Masouros, and Y. Zeng, "ISAC from the sky: UAV trajectory design for joint communication and target localization," *IEEE Trans. Wireless Commun.*, vol. 23, no. 10, pp. 12857–12872, Oct. 2024.
- [17] S. P. Sira et al., "Adaptive waveform design for improved detection of low-RCS targets in heavy sea clutter," *IEEE J. Sel. Topics Signal Process.*, vol. 1, no. 1, pp. 56–66, Jun. 2007.
- [18] C. Wang, X. Song, Q. Fu, L. Cui, and P. Chen, "Design and optimization of PV power supply system for marine buoys," *J. Mar. Sci. Eng.*, vol. 11, no. 9, p. 1808, Sep. 2023.
- [19] L. Xie, X. Cao, J. Xu, and R. Zhang, "UAV-enabled wireless power transfer: A tutorial overview," *IEEE Trans. Green Commun. Netw.*, vol. 5, no. 4, pp. 2042–2064, Dec. 2021.
- [20] M. Li, H. Li, P. Ma, and H. Wang, "Energy maximization for ground nodes in UAV-enabled wireless power transfer systems," *IEEE Internet Things J.*, vol. 10, no. 19, pp. 17096–17109, Oct. 2023.
- [21] W. Feng et al., "Joint 3D trajectory design and time allocation for UAV-enabled wireless power transfer networks," *IEEE Trans. Veh. Technol.*, vol. 69, no. 9, pp. 9265–9278, Sep. 2020.
- [22] G. Park, K. Heo, W. Lee, and K. Lee, "UAV-assisted wireless-powered two-way communications," *IEEE Trans. Intell. Transp. Syst.*, vol. 25, no. 3, pp. 2641–2655, Mar. 2024.
- [23] X. Li et al., "Integrating sensing, communication, and power transfer: From theory to practice," *IEEE Commun. Mag.*, vol. 62, no. 9, pp. 122–127, Sep. 2024.
- [24] Y. Chen, H. Hua, J. Xu, and D. W. K. Ng, "ISAC meets SWIPT: Multi-functional wireless systems integrating sensing, communication, and powering," *IEEE Trans. Wireless Commun.*, vol. 23, no. 8, pp. 8264–8280, Aug. 2024.
- [25] Y. Xu, D. Xu, and S. Song, "Sensing-assisted robust SWIPT for mobile energy harvesting receivers in networked ISAC systems," *IEEE Trans. Wireless Commun.*, vol. 24, no. 3, pp. 2094–2109, Mar. 2025.
- [26] J. Clarke, "Airborne early warning radar," *Proc. IEEE*, vol. 73, no. 2, pp. 312–324, 1985.
- [27] C. A. Balanis, *Antenna Theory: Analysis and Design*. Hoboken, NJ, USA: Wiley, 2015.
- [28] C. Ouyang, Y. Liu, H. Yang, and N. Al-Dhahir, "Integrated sensing and communications: A mutual information-based framework," *IEEE Commun. Mag.*, vol. 61, no. 5, pp. 26–32, May 2023.
- [29] S. Wang, L. Chen, J. Zhou, Y. Chen, K. Han, and C. You, "Unified ISAC Pareto boundary based on mutual information and minimum mean-square error estimation," *IEEE Trans. Commun.*, vol. 72, no. 11, pp. 6783–6795, Nov. 2024.
- [30] J. Li, G. Zhou, T. Gong, and N. Liu, "A framework for mutual information-based MIMO integrated sensing and communication beamforming design," *IEEE Trans. Veh. Technol.*, vol. 73, no. 6, pp. 8352–8366, Jun. 2024.
- [31] X. Zheng, Y. Wu, L. Fan, X. Lei, R. Qingyang Hu, and G. K. Karagiannidis, "Dual-functional UAV-empowered space-air-ground networks: Joint communication and sensing," *IEEE J. Sel. Areas Commun.*, vol. 42, no. 12, pp. 3412–3427, Dec. 2024.
- [32] E. Boshkovska, D. W. K. Ng, N. Zlatanov, and R. Schober, "Practical non-linear energy harvesting model and resource allocation for SWIPT systems," *IEEE Commun. Lett.*, vol. 19, no. 12, pp. 2082–2085, Dec. 2015.
- [33] R. Soulsby, "Simplified calculation of wave orbital velocities," HR Wallingford, Wallingford, U.K., Rep. TR 155, 2006.
- [34] K. Shen and W. Yu, "Fractional programming for communication systems—Part I: Power control and beamforming," *IEEE Trans. Signal Process.*, vol. 66, no. 10, pp. 2616–2630, May 2018.
- [35] B. R. Marks and G. P. Wright, "Technical note—A general inner approximation algorithm for nonconvex mathematical programs," *Oper. Res.*, vol. 26, no. 4, pp. 681–683, Aug. 1978.
- [36] B. T. Polyak, "Convexity of quadratic transformations and its use in control and optimization," *J. Optim. Theory Appl.*, vol. 99, no. 3, pp. 553–583, Dec. 1998.
- [37] H. V. Nguyen, V.-D. Nguyen, O. A. Dobre, Y. Wu, and O.-S. Shin, "Joint antenna array mode selection and user assignment for full-duplex MU-MISO systems," *IEEE Trans. Wireless Commun.*, vol. 18, no. 6, pp. 2946–2963, Jun. 2019.
- [38] Y. Pan et al., "Cooperative trajectory planning and resource allocation for UAV-enabled integrated sensing and communication systems," *IEEE Trans. Veh. Technol.*, vol. 73, no. 5, pp. 6502–6516, May 2024.
- [39] T. P. Boyer et al., "World ocean database 2013," NOAA Atlas NESDIS 72, Sydney Levitus, Silver Spring, MD, USA, 2013.
- [40] K. Meng, Q. Wu, S. Ma, W. Chen, K. Wang, and J. Li, "Throughput maximization for UAV-enabled integrated periodic sensing and communication," *IEEE Trans. Wireless Commun.*, vol. 22, no. 1, pp. 671–687, Jan. 2023.